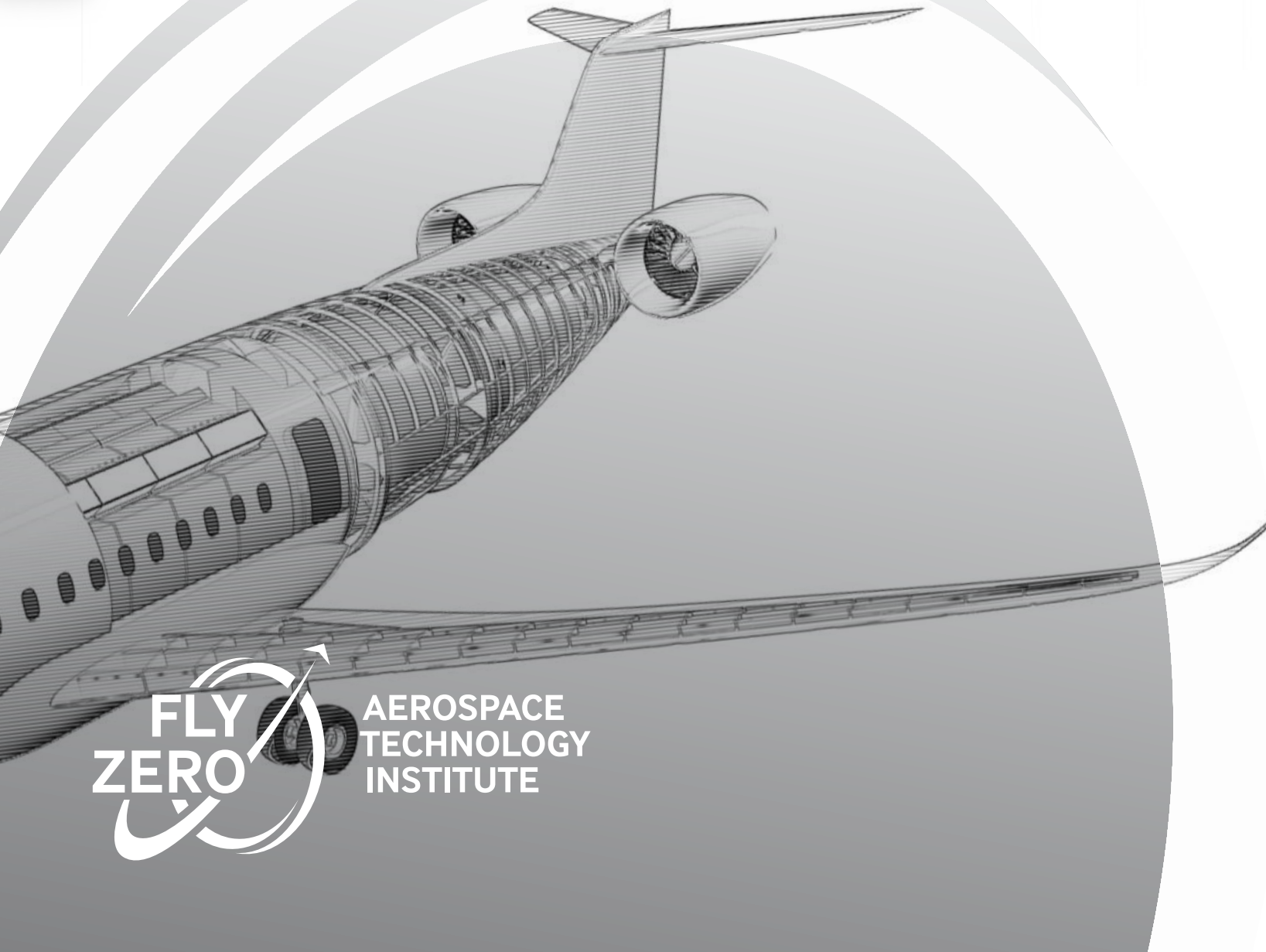




AERODYNAMIC STRUCTURES

Roadmap Report



FZO-AIR-COM-0016

Published March 2022

CONTENTS

OVERVIEW: AERODYNAMIC STRUCTURES

KEY FINDINGS: AERODYNAMIC STRUCTURES

AIRCRAFT CONFIGURATION & TECHNOLOGY SELECTION

TECHNOLOGY TARGETS

AERODYNAMIC STRUCTURES ROADMAP

TECHNOLOGY ASSESSMENT

TECHNOLOGY STAIRCASE

CONCLUSION: AERODYNAMIC STRUCTURES

RELATED FLYZERO FURTHER READING

ABOUT FLYZERO

ACKNOWLEDGEMENTS

KEY & LIST OF ABBREVIATIONS

3 **Key**

4

5 **Technology Development** Development required to bring roadmap technology to TRL6.

6

7 **Technology Mature** Transition to further integration and production readiness.

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14 **Essential Enabler** Supporting activity or infrastructure to enable technology development

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List of Abbreviations

A/C – Aircraft

BLI – Boundary Layer Ingestion

CG – Centre of Gravity

CMC – Ceramic Matrix Composites

EMI – Electro-Magnetic Interference

HAR – High Aspect Ratio

HTP – Horizontal Tailplane

LE – Leading Edge

L.F.– Laminar Flow

LH₂ – Liquid Hydrogen

MDO – Multi-Disciplinary (MD) Optimisation

MRO – Maintenance, Repair & Overhaul

NDT – Non-Destructive Testing

Opt. – Optimisation

SFC – Specific Fuel Consumption

SMA – Shape Memory Alloys

SMPC – Shape Memory Polymer Composites

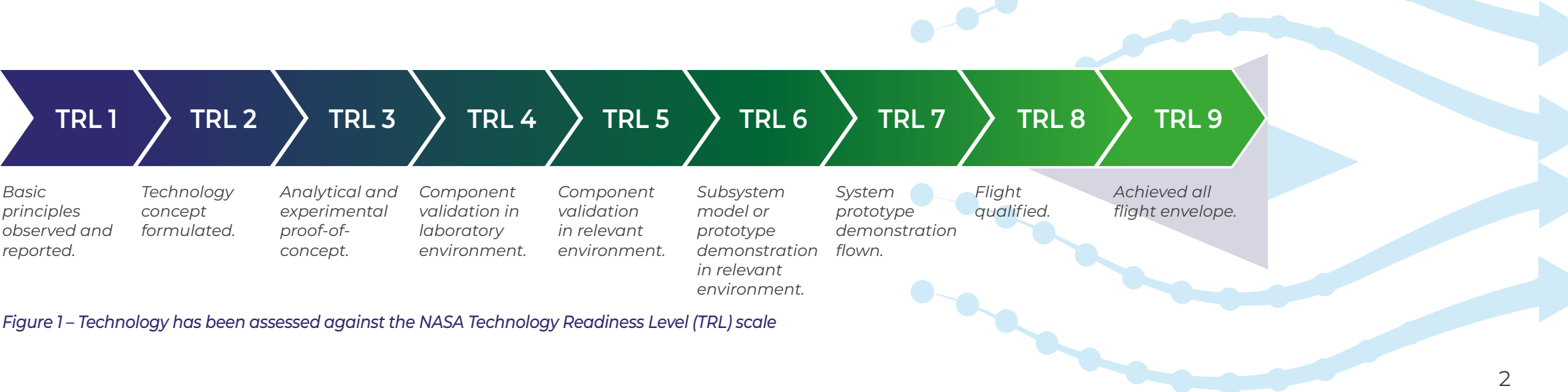
TE – Trailing Edge

UERF – Uncontained Engine Rotor Failure

VAN – Variable Area Nozzle

Volumetric energy density – available energy per unit volume

VTP – Vertical Tailplane



OVERVIEW: AERODYNAMIC STRUCTURES

This Aerodynamic Structures Roadmap, encompassing aerodynamics, loads, aeroelastics, low-weight structural design, airframe optimisation, manufacturing, and assembly, presents a selection of high potential technologies that deliver aerodynamic performance improvements, fuel burn improvements and airframe weight reductions.

Aircraft performance improvements and weight savings are a means to reduce fuel burn, and consequently, operating costs and emissions. FlyZero concludes that liquid hydrogen fuelled aircraft are feasible, but aviation will continue to need such improvements for four reasons:

- To reduce volume and mass of hydrogen needed for a given mission
- To minimise operational costs and non-CO₂ emissions
- To limit demand for green energy for production of sustainable aviation fuel (SAF) and liquid hydrogen
- To limit CO₂ emissions of kerosene and SAF aircraft during transition to a liquid hydrogen fleet

FlyZero has identified critical Aerodynamic Structures technologies in six clusters of swimlanes that require immediate funding, research and development to maturity to enable zero-carbon aircraft of the future.



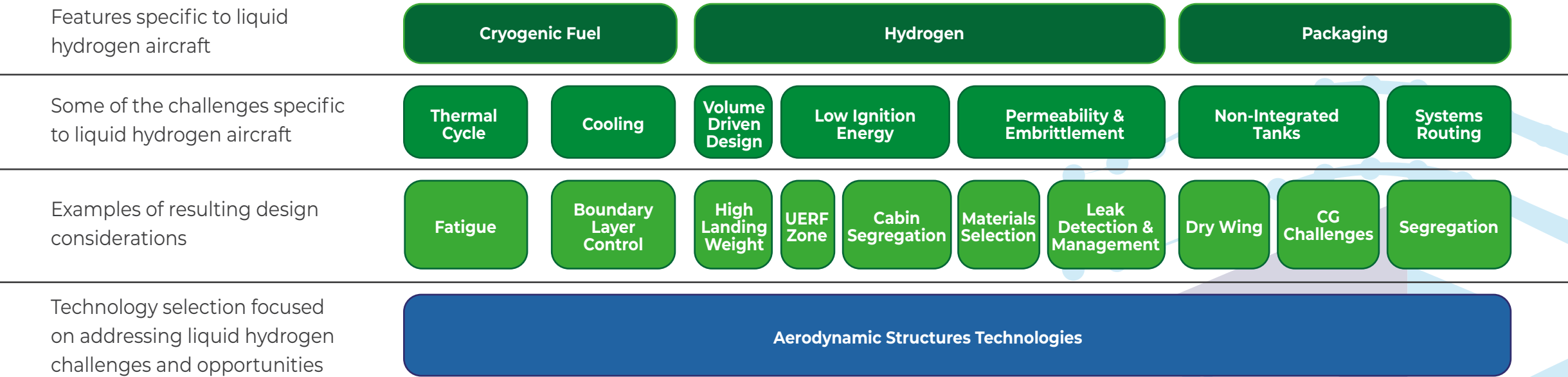
Note: Due to the large number of technologies identified, only a selection of high potential technologies are shown in this roadmap. A complete list with additional detail is provided in the FlyZero Aerodynamic Structures Technical Report.

Note: Direct Specific Fuel Consumption savings from propulsion system improvements are not covered as part of the Performance Improvements cluster of this roadmap.

KEY FINDINGS: AERODYNAMIC STRUCTURES

While some of the technologies in the Aerodynamic Structures Roadmap are unique to a liquid hydrogen aircraft, many are also applicable to kerosene and sustainable aviation fuelled aircraft. Therefore, these common technologies will have benefits regardless of the next generation aircraft fuel choice.

To focus the technology selection, features specific to liquid hydrogen were identified and the resulting design considerations explored. For example, cryogenic hydrogen results in spherical or cylindrical tanks not integrated in the wing structure. This, and the low volumetric energy density, may lead to a relatively wide diameter fuselage or external tanks, increasing airframe drag. Therefore, aerodynamic improvements such as laminar flow are selected for the roadmap.



AIRCRAFT CONFIGURATION & TECHNOLOGY SELECTION



Aircraft Configuration

Aerodynamic Structures technologies are inherently linked with aircraft configuration. Some novel configurations are included in the roadmap and others are discussed in the Aerodynamic Structures Technical Report.

Although many novel configurations are well researched, the challenge to design, build and certify them is recognised in the roadmap with longer timescales.



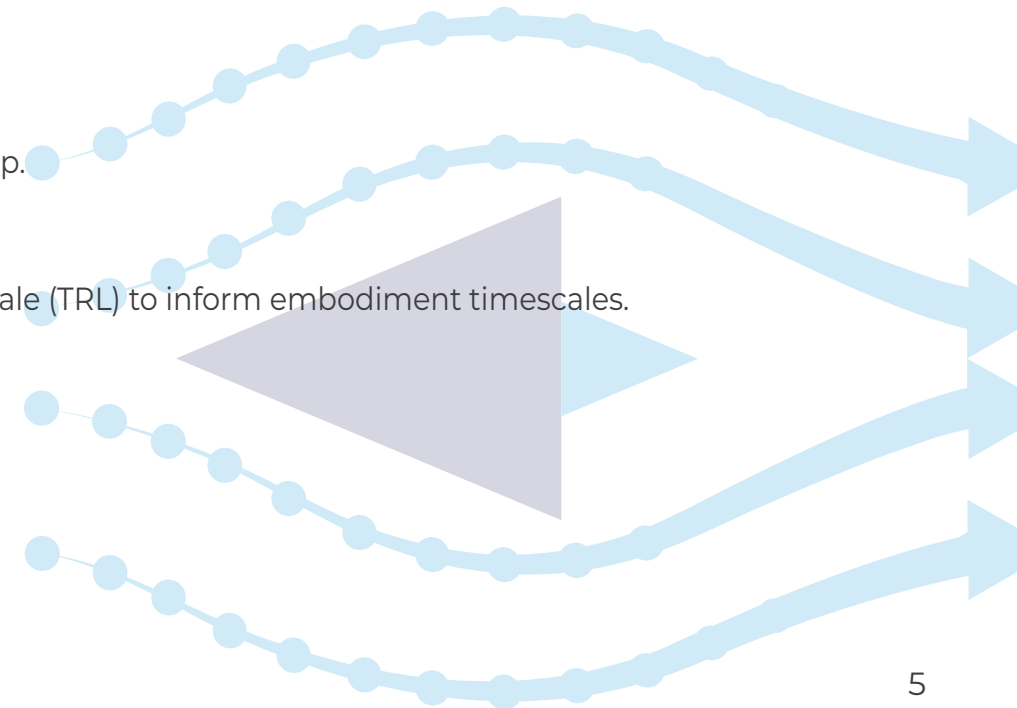
Technology Selection

The purpose of the roadmap is to accelerate technology development to TRL9 for aircraft entering into service in the 2030s. Technologies were selected based on maturity.

If a technology is already TRL9+ in 2022 it was not selected for the roadmap.


If a technology was not expected to reach TRL6 by 2030 it was not selected for the roadmap.

Technology maturity has been assessed against the NASA Technology Readiness Levels scale (TRL) to inform embodiment timescales.



TECHNOLOGY TARGETS

For the Aerodynamic Structures Roadmap, a set of ambitious but achievable aircraft improvement targets have been set which should be targeted by industry over the time periods shown in the table below.



| | 2025 | 2030 | 2035 | 2050 |
|--|------|------|------|------|
| Weight Saving | 20% | 25% | 30% | 40% |
| Performance/SFC/CO ₂ Emissions Improvements | 10% | 15% | 20% | 30% |

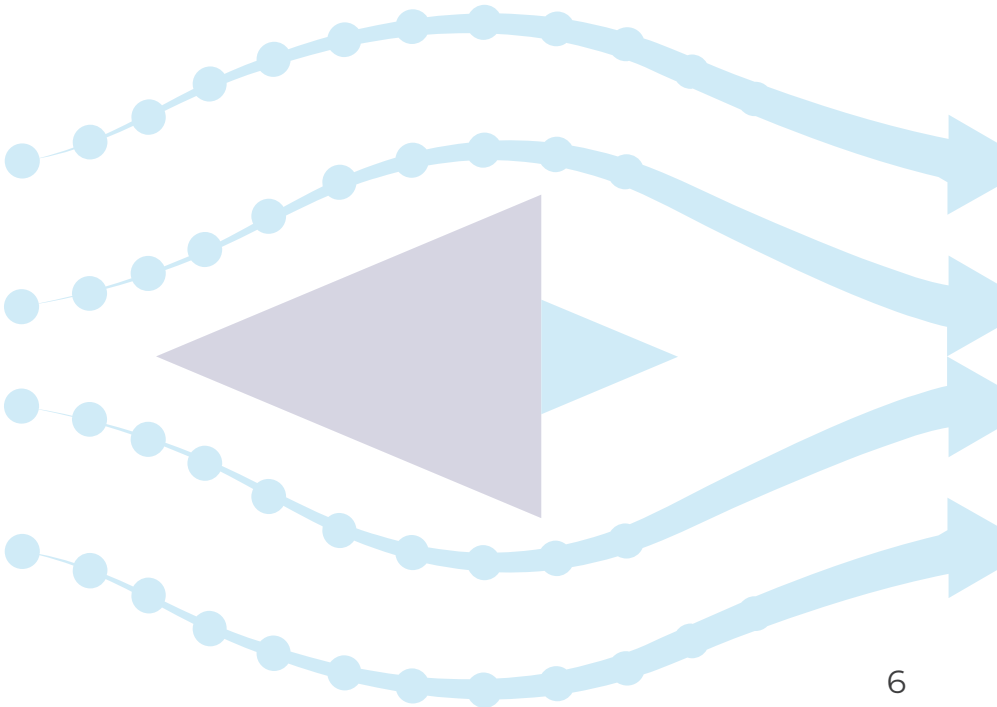
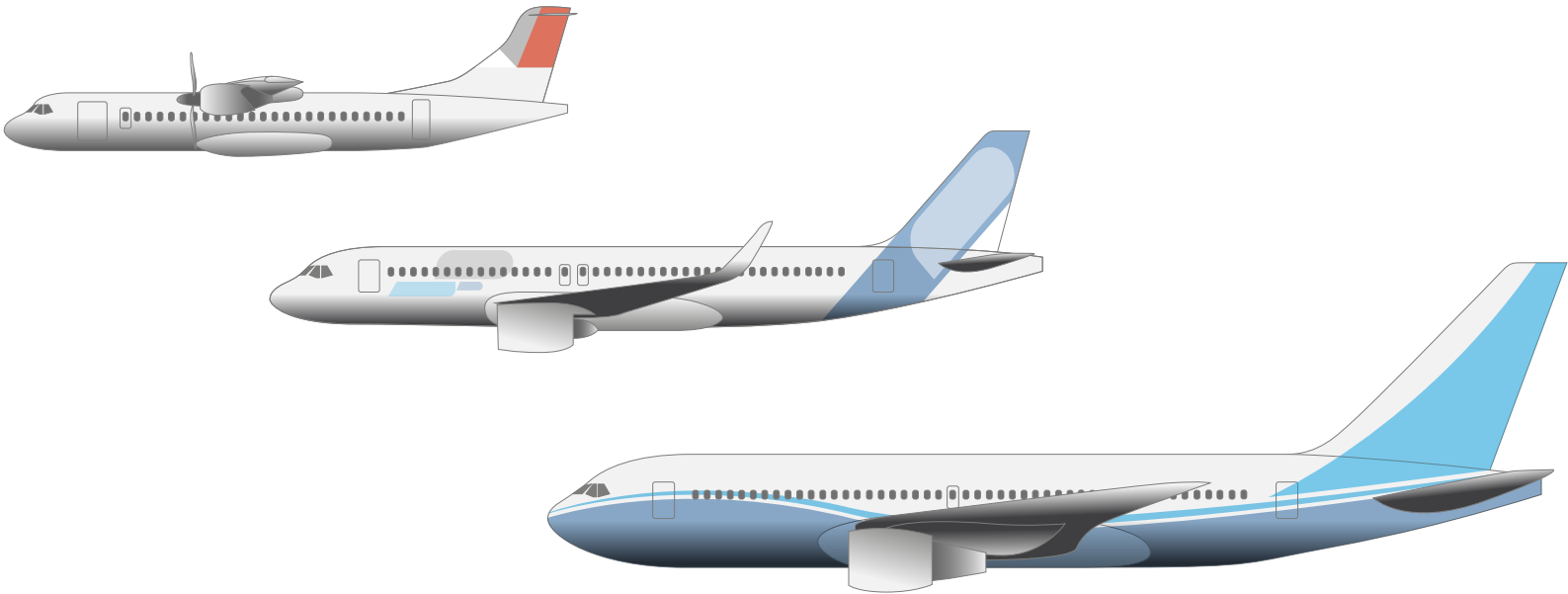
These targets are improvements over reference aircraft in three sectors: Regional, Narrowbody & Midsize.

Target values have been selected based on reviews of other technology roadmaps, literature, and key stakeholder consultations.

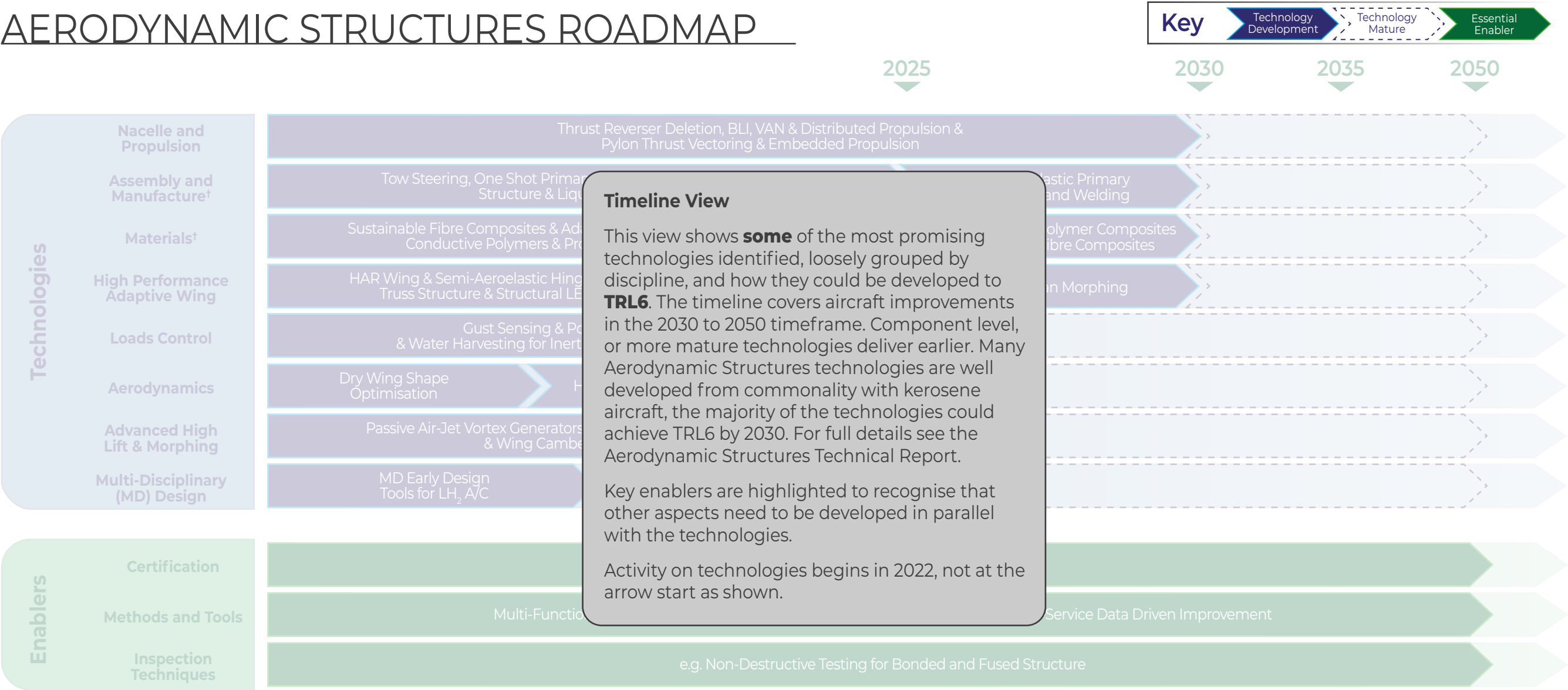
Not all of the technologies presented within the roadmap will be implemented in aircraft of the future, but all of them should be targeted for research and development as they represent an opportunity to take a step towards achieving these targets.

In the context of reducing aircraft emissions and enabling early adoption of zero-carbon aircraft these targets should be pursued at pace.

Figure 3 - Percentage improvement targets compared with reference ATR72-600 (below top), Airbus A320neo (below middle) or Boeing 767-200ER (below bottom) aircraft

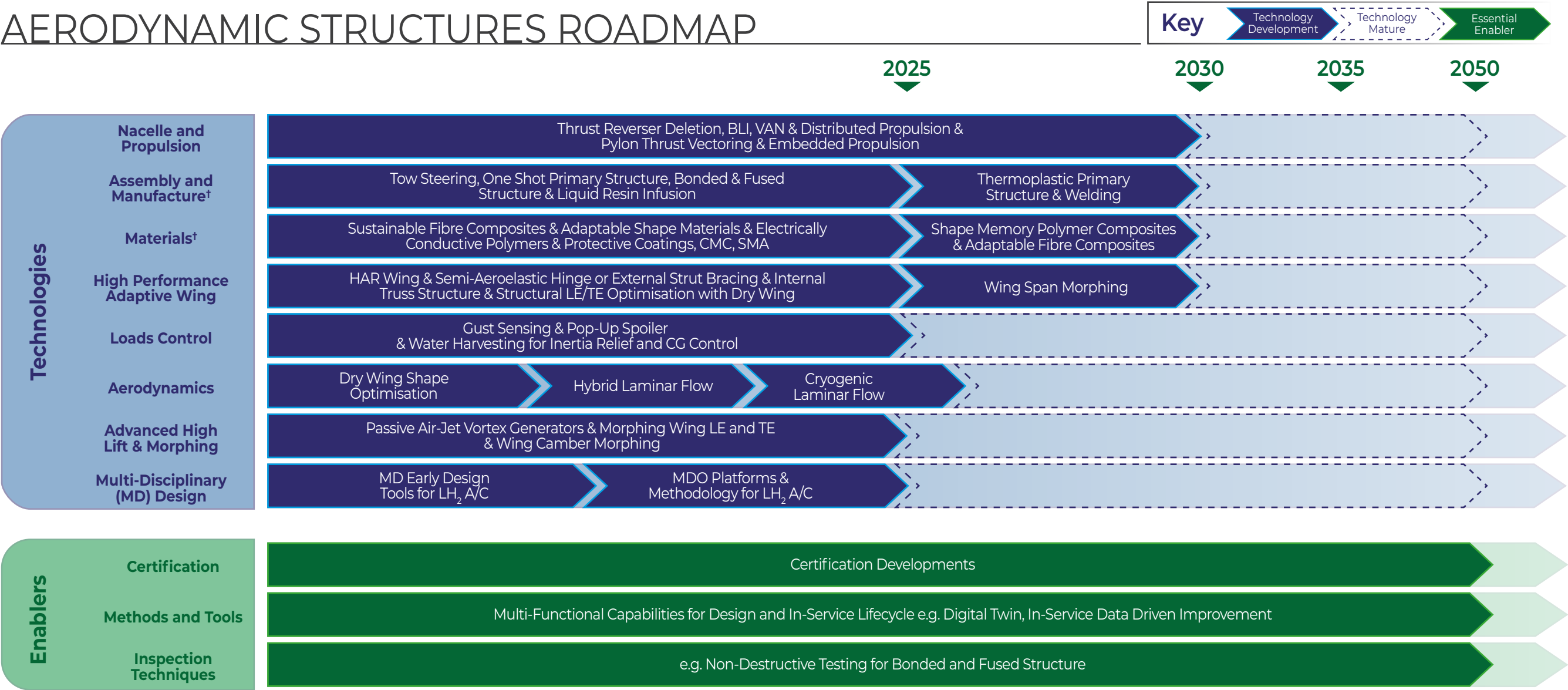


AERODYNAMIC STRUCTURES ROADMAP



† The focus in this roadmap for the materials and manufacturing clusters has been on composites given their high potential for weight saving however metallics are still a primary choice for airframe structures and technology development in this area should continue apace.

AERODYNAMIC STRUCTURES ROADMAP



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TECHNOLOGY ASSESSMENT

A Technology Assessment for cost, weight and performance has been performed based on literature reviews, academia & industry engagement, and engineering judgement to show the potential estimated benefits of technologies at the swim lane level.

Technology indicators show likely trends to be achieved at aircraft level by 2030-2050, but these may change as technology is further developed and matures. Technology indicators may change depending on the application of the technology on aircraft or advances in key enablers.

The definition of each technology indicator is presented below.

Cost

Cost refers to Non-Recurring Cost (NRC) related to the manufacture of the aircraft or components.

Buy-to-Fly Ratio or recurring costs are not covered by this indicator.

Cost indicators reflect today's cost to enable technology deployment, but costs should reduce with maturity and the introduction of more advances in materials and manufacturing technologies.

Weight

Weight refers to total airframe weight, not whole aircraft weight reduction.

The technology weight indicator trend reflects that while some technologies may be heavy or lighter individually, their integration on to an aircraft may result in an overall weight increase or reduction by addition to or replacement of other technologies.

Performance

Performance refers to aircraft level operational performance.

This includes operational costs, lift to drag ratio & fuel efficiency of the aircraft.

It does not include any performance gains due to weight reduction improvements.

TECHNOLOGY ASSESSMENT

| | | Cost | Weight | Performance | % Benefit |
|--|--|------|--------|-------------|----------------------|
| Advanced and Novel/ Adaptable Materials | Ceramic Matrix Composites | - | + | + | 1-3% SFC improvement |
| | Shape Memory Alloys | - | 0 | ++ | |
| | Shape Memory Polymer Composites | - | 0 | ++ | |
| | Adaptable Fibre Composites | - | 0 | ++ | |
| | Electrically Conductive Polymers | - | 0 | ++ | |
| Laminar Flow | Natural Laminar Flow | - | 0 | ++ | 1-3% SFC improvement |
| | Hybrid/ Natural Laminar Nacelle Technology | - | 0 | ++ | |
| | Hybrid Laminar Flow | - | 0 | ++ | |
| | Cryogenic Laminar Flow | - | 0 | ++ | |
| Lift to Drag Ratio Improvement | Riblets (Non-Mechanically Fastened) | - | 0 | ++ | 1-3% SFC improvement |
| | Dry Wing Shape Optimisation | - | 0 | ++ | |
| Advanced Moveables and High Lift | Passive Air-Jet Vortex Generators | - | 0 | ++ | 1-3% SFC improvement |
| | Coanda Blowing | 0 | + | + | |
| | Multi-Functional Trailing Edge | 0 | 0 | + | |

Technology Assessment View

This view shows the expected percentage benefit of **some** of the most promising technologies identified. A colour scale gives a qualitative assessment of each technology against the technology indicators.

For full details see the Aerodynamic Structures Technical Report.

Combining the percentage benefit with the colour scale provides a view of the potential for each technology.

All the technologies merit further work but some may offer improvements at lower cost, or may offer more significant benefits.

Key

- Negative trend - high
- Negative trend - medium
- 0 No change/ marginal change
- +
- ++ Positive trend - high benefit
- n/a Not applicable

TECHNOLOGY ASSESSMENT

| | | Cost | Weight | Performance | % Benefit |
|--|---|------|--------|-------------|---|
| Advanced and Novel/ Adaptable Materials | Ceramic Matrix Composites | - | + | + | 1-5% weight reduction (in addition to metallic to composite conversion) improvement |
| | Shape Memory Alloys | - | 0 | ++ | |
| | Shape Memory Polymer Composites | - | 0 | ++ | |
| | Adaptable Fibre Composites | - | + | ++ | |
| | Electrically Conductive Polymers | 0 | + | n/a | |
| Laminar Flow | Natural Laminar Flow | - | 0 | ++ | 10% SFC improvement |
| | Hybrid/Natural Laminar Nacelle Technology | - | 0 | + | |
| | Hybrid Laminar Flow | -- | - | ++ | |
| | Cryogenic Laminar Flow | - | - | ++ | |
| Lift to Drag Ratio Improvement | Riblets (Non-Mechanically Fastened) | - | 0 | + | 3-5% SFC improvement |
| | Dry Wing Shape Optimisation | 0 | 0 | ++ | |
| Advanced Moveables and High Lift | Passive Air-Jet Vortex Generators | 0 | + | + | 1-3% SFC improvement |
| | Coanda Blowing | 0 | + | + | |
| | Multi-Functional Trailing Edge | 0 | 0 | + | |

Key

--

Negative trend - high

-

Negative trend - medium

0

No change/ marginal change

+

Positive trend - medium benefit

++

Positive trend - high benefit

n/a

Not applicable

TECHNOLOGY ASSESSMENT

| | | Cost | Weight | Performance | % Benefit |
|-------------------------------|---|------|--------|-------------|-----------------------|
| Morphing Structures | Wing Camber Morphing | - | - | + | 2-8% SFC improvement |
| | Adaptive Wing Tip | - | - | + | |
| | Wing Span Morphing | -- | - | + | |
| | Morphing Nacelle Inlets | - | 0 | + | |
| | Control Surface Morphing (on HTP/VTP) | 0 | - | + | |
| Loads and Aeroelastic Control | Semi-Aeroelastic Hinge | - | ++ | ++ | 1-5% SFC improvement |
| | Gust Sensing Devices | - | + | n/a | |
| | Aeroelastic Tailoring (Tow Steered) | - | ++ | + | |
| Novel Structural Design | Truss Internal Wing Structure | - | ++ | n/a | 5% SFC improvement |
| | Folding Wing Tip | - | - | + | |
| | Topology Optimisation/Generative Design | 0 | ++ | 0 | |
| Novel Wing Architectures | Wing External Strut Bracing | -- | + | ++ | 5-10% SFC improvement |

Key

--

Negative trend - high

-

Negative trend - medium

0

No change/ marginal change

+

Positive trend - medium benefit

++

Positive trend - high benefit

n/a

Not applicable

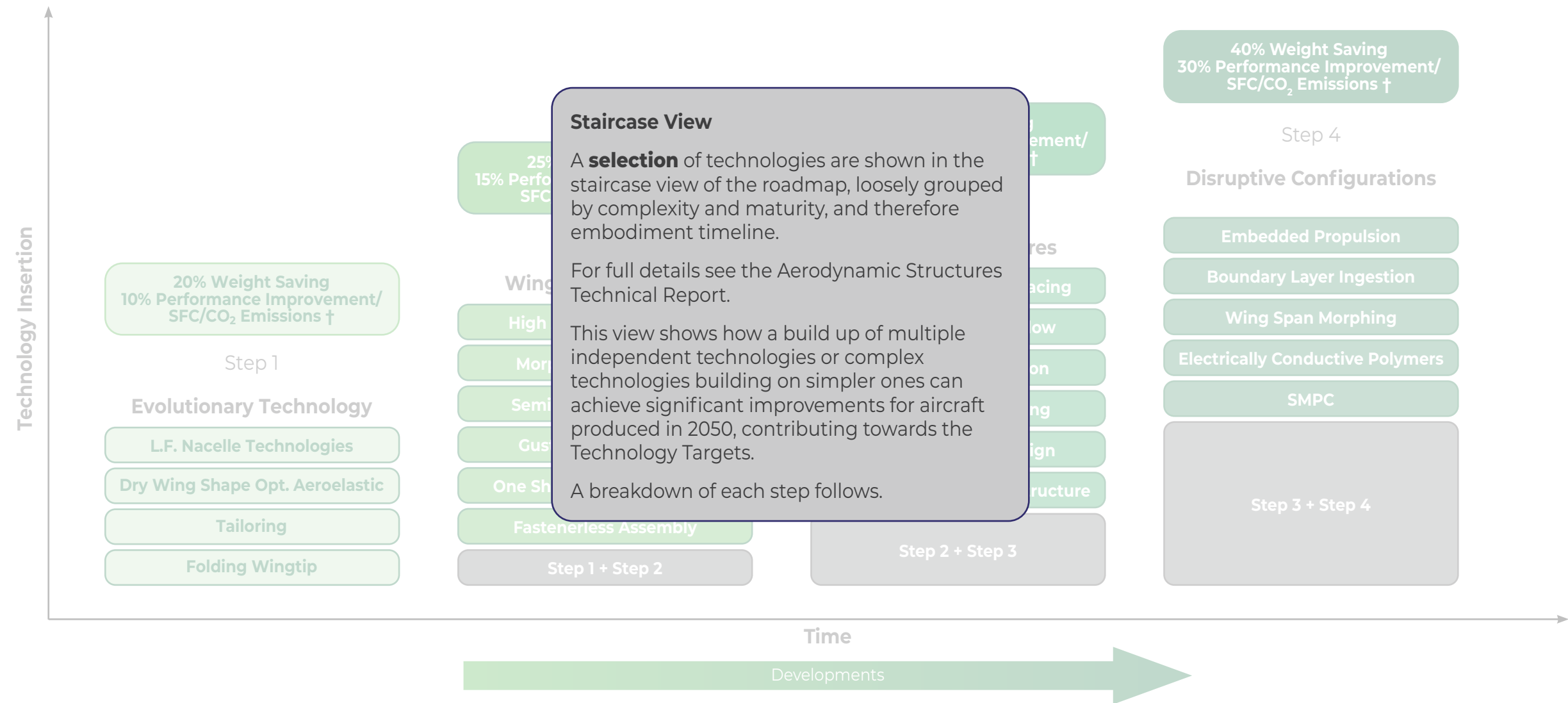
TECHNOLOGY ASSESSMENT

| | | Cost | Weight | Performance | % Benefit |
|------------------------------------|--|------|--------|-------------|-------------------------|
| Novel Nacelle and Pylon Design | Distributed Propulsion | 0 | 0 | + | 5% SFC improvement |
| | Nacelle Technologies - Thrust Reverser Deletion, BLI and VAN | + | + | + | |
| | Embedded Propulsion | -- | + | + | |
| | Moveable Pylon for Thrust Vectoring | - | 0 | + | |
| Additive Manufacturing | Additive Manufacturing (Metallic) | + | ++ | + | 10-20% weight reduction |
| | Additive Manufacturing (Composite) | - | + | + | |
| Manufacturing and Assembly Process | Liquid Resin Infusion | ++ | + | + | 5-10% weight reduction |
| | Bonded & Fused Primary Structure with NDT | ++ | ++ | + | |
| | One Shot Primary Structure | + | + | 0 | |
| | Thermoplastic Primary Structure | ++ | ++ | + | |
| Multi-Disciplinary (MD) Design | Platform Development | - | ++ | ++ | 1-3% SFC improvement |
| | Methodology Development | ++ | ++ | ++ | |

Key

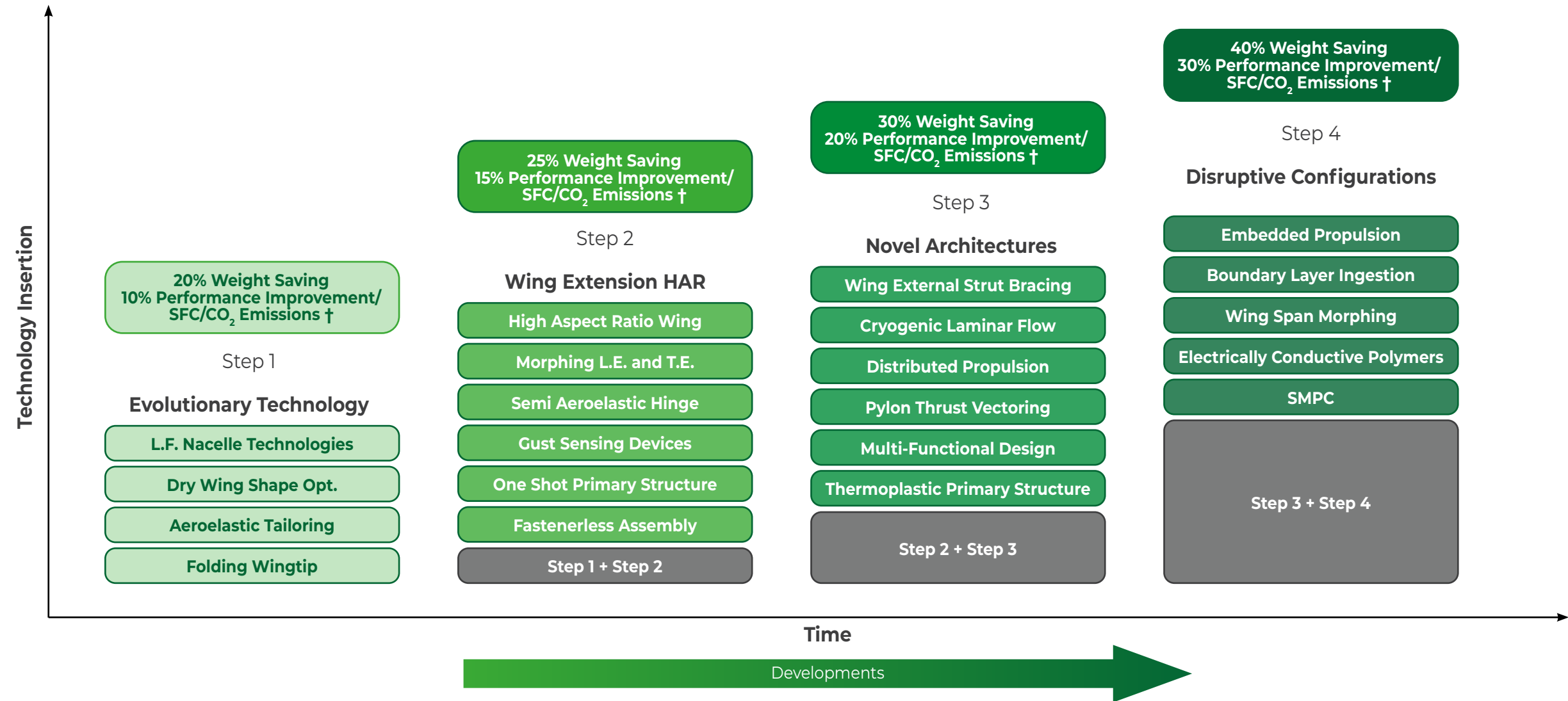
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- n/a Not applicable

TECHNOLOGY STAIRCASE



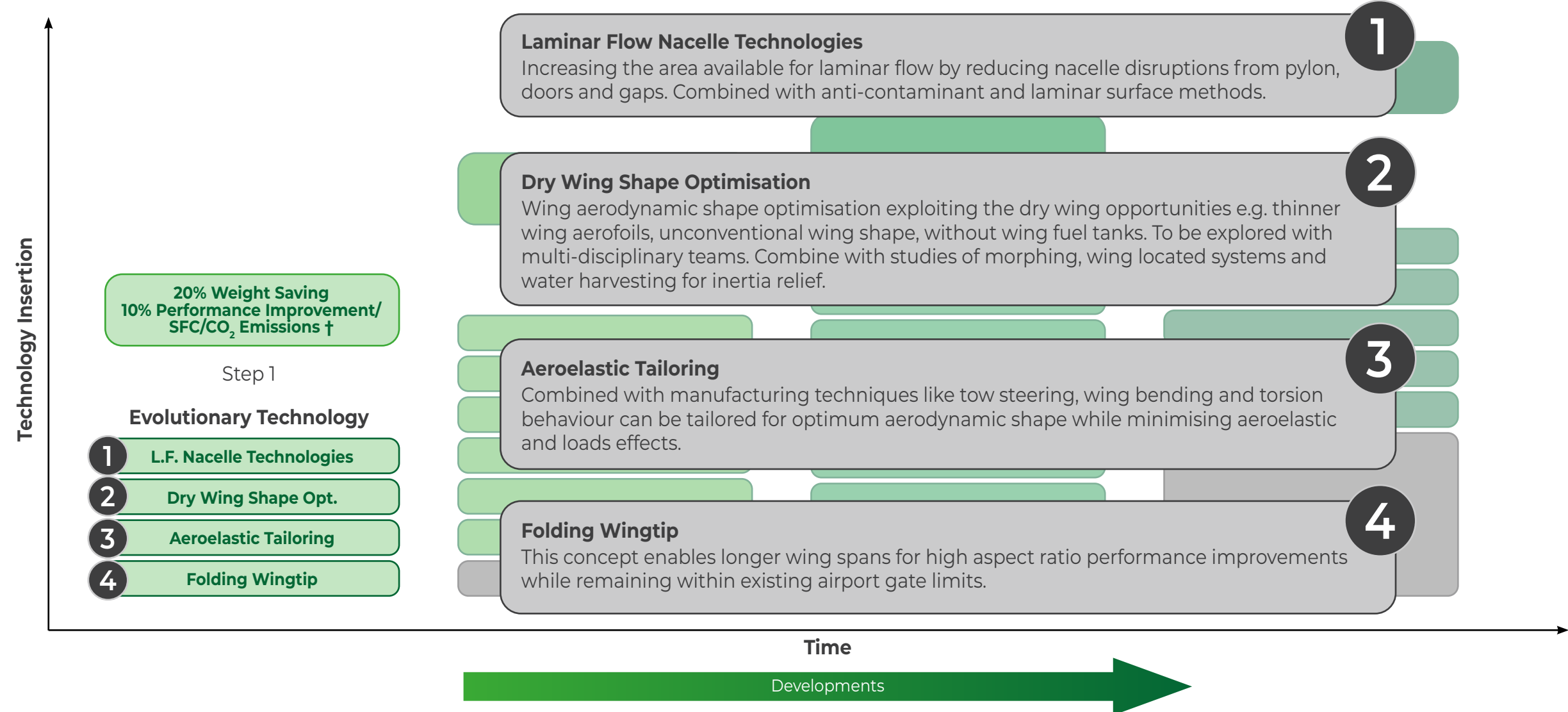
† Potential total gains are a combination of technologies which is not limited to those shown here nor the sum of all

TECHNOLOGY STAIRCASE



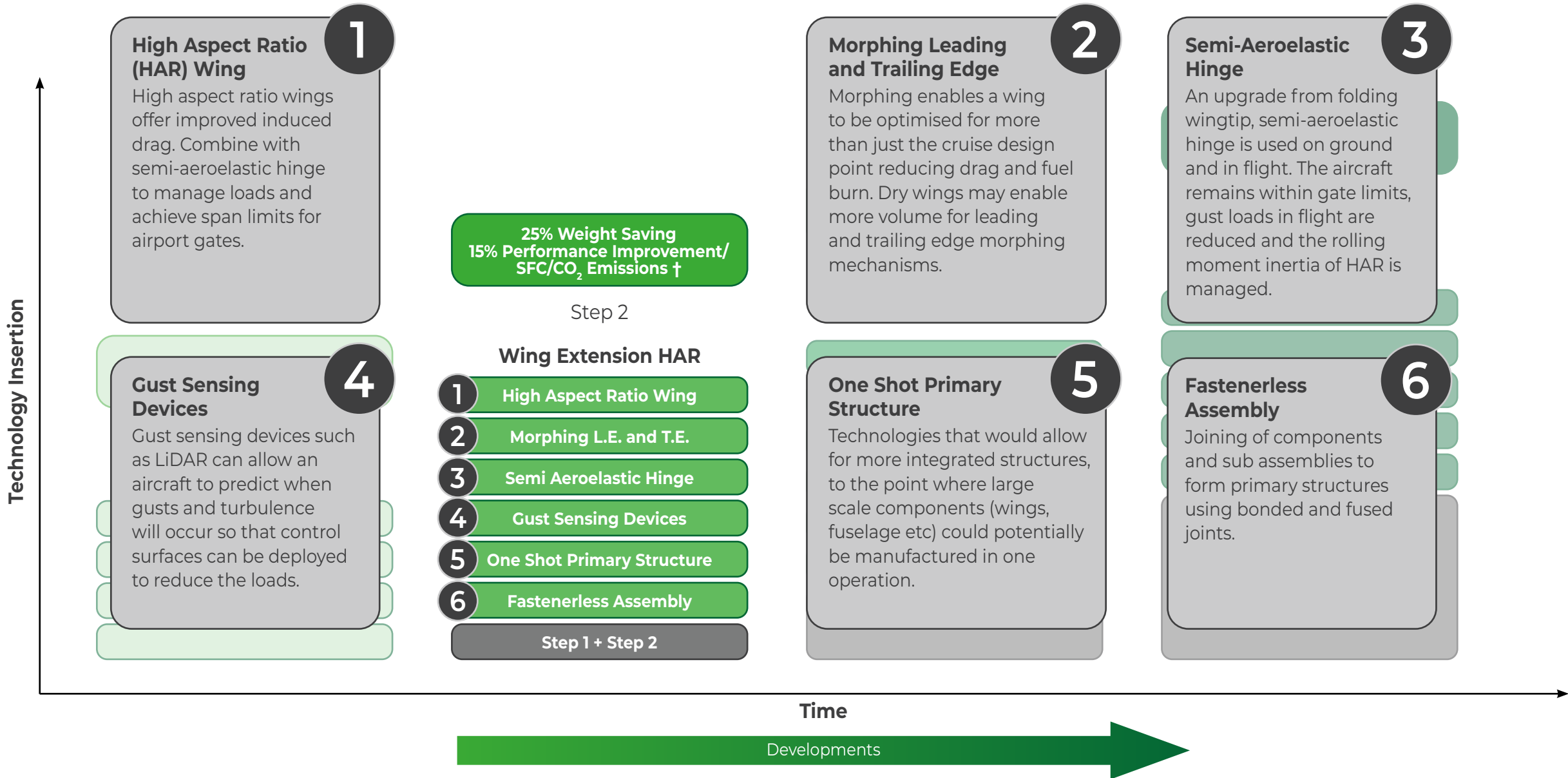
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TECHNOLOGY STAIRCASE - STEP 1



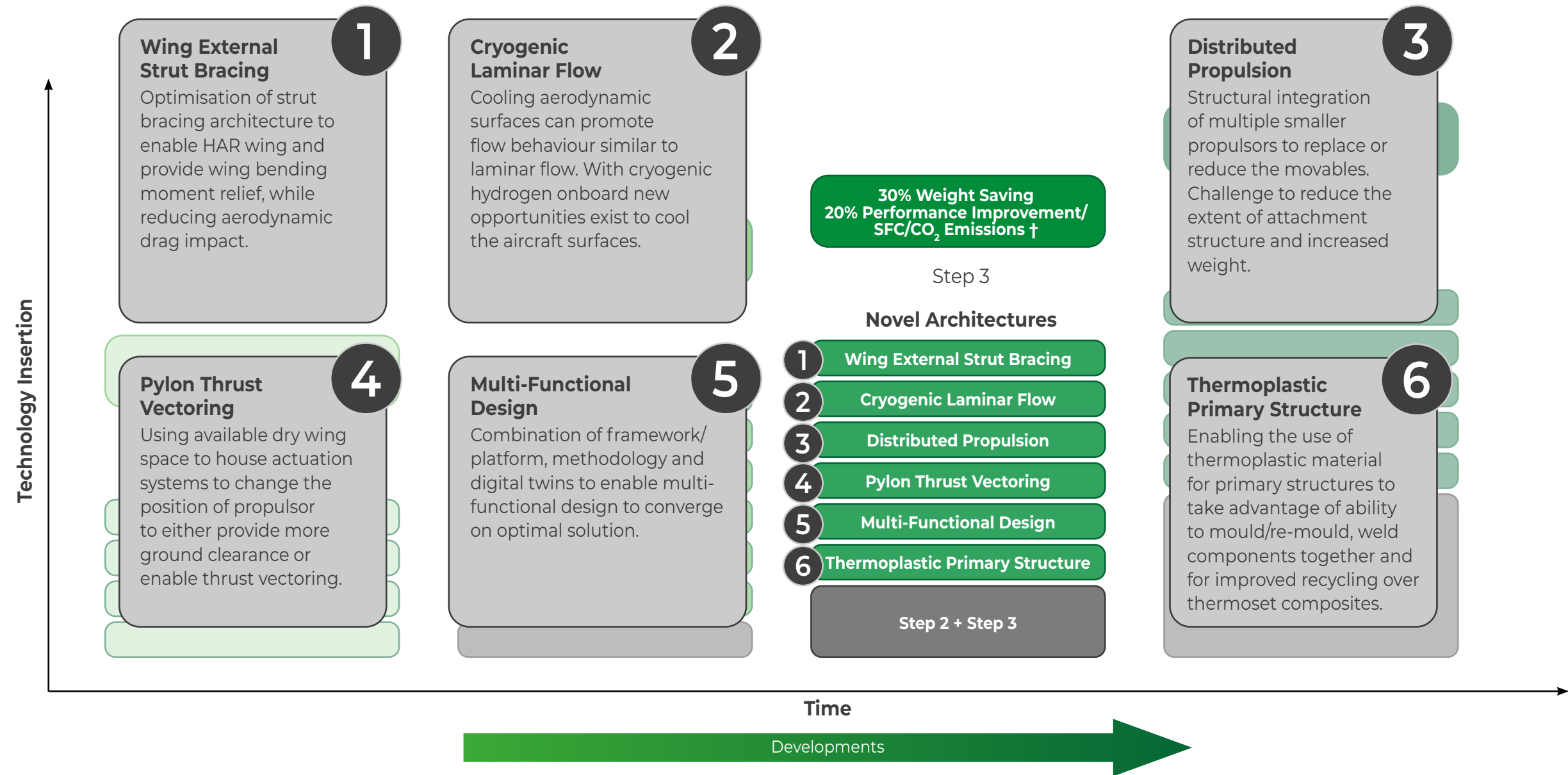
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TECHNOLOGY STAIRCASE - STEP 2



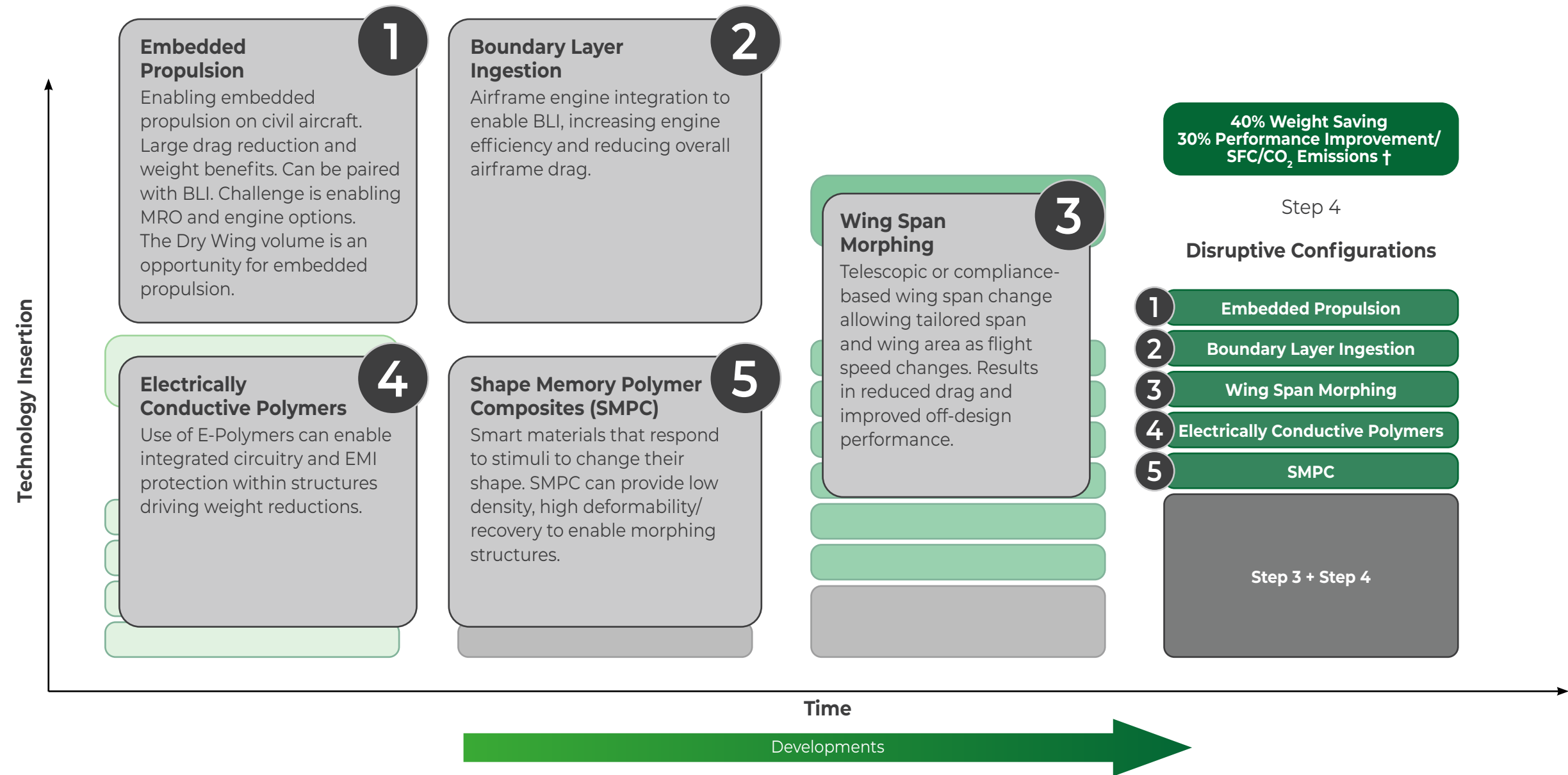
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TECHNOLOGY STAIRCASE - STEP 3



† Potential total gains are a combination of technologies which is not limited to those shown here nor the sum of all

TECHNOLOGY STAIRCASE - STEP 4



† Potential total gains are a combination of technologies which is not limited to those shown here nor the sum of all

CONCLUSION: AERODYNAMIC STRUCTURES

The Aerodynamic Structures Roadmap, encompassing aerodynamics, loads, aeroelastics, low-weight structural design, airframe optimisation, manufacturing, and assembly, presents a selection of high potential technologies that deliver aerodynamic performance improvements, fuel burn improvements and airframe weight reductions.

The Aerodynamic Structures Technical Report includes all the identified technologies with potential to deliver meaningful airframe improvements in the ambitious timescales of commercial entry into service in the 2030s, meriting investment and development.

Research continuously discovers new ideas and technologies. New technologies that contribute to significant aerodynamic performance improvements, fuel burn improvements or airframe weight savings by 2030 should also be proposed for investment and accelerated development.

One of the key messages from FlyZero is that understanding and development of these key technologies must be accelerated to deliver a viable zero-carbon aircraft product in the 2030s.



RELATED FLYZERO FURTHER READING

The ATI FlyZero project developed its technology roadmaps through a combination of broad industry consultation and assessment of technologies by experts. Technology assessment was carried out both by the FlyZero team and by approximately 50 industrial and academic organisations that partnered with FlyZero to support delivery. During the project, FlyZero developed three concept aircraft and used this exercise to gain a deep understanding of requirements and challenges for systems and technologies, which have been reflected in the roadmaps. Further detail of these technologies and developments can be found in the following reports, available to download from ati.org.uk:

FlyZero

**Zero-Carbon Emission Aircraft Concepts**
Report
Ref. FZO-AIN-REP-0007

**Technology Roadmaps**
Report
Ref. FZO-IST-MAP-0012

**Workforce to Deliver Liquid Hydrogen Powered Aircraft**
Report
Ref. FZO-IST-PPL-0053

Hydrogen Aircraft

**Aerodynamic Structures**
Technical Report
Ref. FZO-AIR-REP-014
Roadmap
Ref. FZO-AIR-MAP-0015
Roadmap Report
Ref. FZO-AIR-COM-0016
Capability Report
Ref. FZO-AIR-CAP-0066

**Thermal Management**
Technical Report
Ref. FZO-PPN-REP-017
Roadmap
Ref. FZO-PPN-MAP-0018
Roadmap Report
Ref. FZO-PPN-COM-0019
Capability Report
Ref. FZO-PPN-CAP-0067

**Hydrogen Gas Turbines & Thrust Generation**
Gas Turbine Technical Report
Ref. FZO-PPN-REP-020
Thrust Devices Technical Report
Ref. FZO-PPN-REP-021
Roadmap
Ref. FZO-PPN-MAP-0022
Roadmap Report
Ref. FZO-PPN-COM-0023
Capability Report
Ref. FZO-PPN-CAP-0068

**Electrical Propulsion System**
Technical Report
Ref. FZO-PPN-REP-0028
Roadmap
Ref. FZO-PPN-MAP-0029
Roadmap Report
Ref. FZO-PPN-COM-0030
Capability Report
Ref. FZO-PPN-CAP-0070

**Fuel Cells**
Technical Report
Ref. FZO-PPN-REP-0031
Roadmap
Ref. FZO-PPN-MAP-0032
Roadmap Report
Ref. FZO-PPN-COM-0033
Capability Report
Ref. FZO-PPN-CAP-0071

**Cryogenic Hydrogen Fuel System & Storage**
Fuel System Technical Report
Ref. FZO-PPN-REP-024
Fuel Storage Technical Report
Ref. FZO-PPN-REP-025
Roadmap
Ref. FZO-PPN-MAP-0026
Roadmap Report
Ref. FZO-PPN-COM-0027
Capability Report
Ref. FZO-PPN-CAP-0069

Cross-Cutting

**Aircraft Systems**
Ref. FZO-AIR-POS-0013

**Airports, Airlines, Airspace - Operations & Hydrogen Infrastructure**
Ref. FZO-CST-POS-0035

**Advanced Materials**
Ref. FZO-IST-POS-0036

**Lifecycle Impact**
Ref. FZO-STY-POS-0034

**Sustainable Cabin Design**
Ref. FZO-AIR-POS-0039

**Compressed Design and Validation - Culture and Digital Tools**
Ref. FZO-IST-POS-0038

**Advanced Manufacturing**
Ref. FZO-IST-POS-0037


ABOUT FLYZERO

Led by the Aerospace Technology Institute and backed by the UK government, FlyZero began in early 2021 as an intensive research project investigating zero-carbon emission commercial flight. This independent study has brought together experts from across the UK to assess the design challenges, manufacturing demands, operational requirements and market opportunity of potential zero-carbon emission aircraft concepts.

FlyZero has concluded that green liquid hydrogen is the most viable zero-carbon emission fuel with the potential to scale to larger aircraft utilising fuel cell, gas turbine and hybrid systems. This has guided the focus, conclusions and recommendations of the project.

This report forms part of a suite of FlyZero outputs which will help shape the future of global aviation with the intention of gearing up the UK to stand at the forefront of sustainable flight in design, manufacture, technology and skills for years to come. To discover more and download the FlyZero reports, visit ati.org.uk

ACKNOWLEDGEMENTS

| | |
|--|--|
| <p>Lead authors</p> <p>Stephen Phillips Structural Design Lead</p> <p>Anna Calder Flight Physics Lead</p> <p>Kuheli Sahu Structural Design Lead</p> <p>Co-authors</p> <p>Adil Dafa'Alla Vijay Sahadevan Paul Kealy Peggy Smith Steven Munn</p> | <p>FlyZero contributing companies</p> <p>Airbus, Belcan, Capgemini, easyJet, Eaton, GE Aviation, GKN Aerospace, High Value Manufacturing Catapult (MTC), Mott MacDonald, NATS, Reaction Engines, Rolls-Royce, Spirit AeroSystems.</p> <p>These roadmaps have been developed with a view to accelerate zero-carbon technology development and maximise the potential future value for the UK. They are unconstrained by the availability of funding.</p> <div><p>Department for Business, Energy & Industrial Strategy</p></div> |
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FlyZero was funded by the Department for Business, Energy and Industrial Strategy.

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AERODYNAMIC STRUCTURES

Roadmap Report

